

# Resistencia de Materiales 6-1

Monday, April 15, 2019 9:30 AM

Problemas varios:

## EXAMPLE 1.11

If the wood joint in Fig. 1–23a has a width of 150 mm, determine the average shear stress developed along shear planes  $a-a$  and  $b-b$ . For each plane, represent the state of stress on an element of the material.

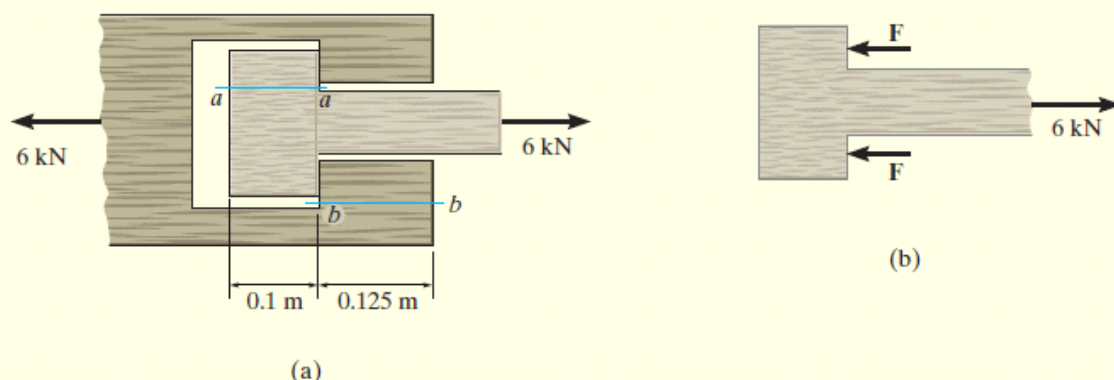


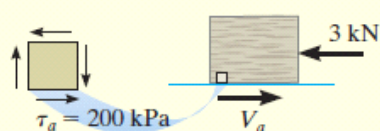
Fig. 1–23

### SOLUTION

**Internal Loadings.** Referring to the free-body diagram of the member, Fig. 1–23b,

$$\pm \rightarrow \Sigma F_x = 0; \quad 6 \text{ kN} - F - F = 0 \quad F = 3 \text{ kN}$$

Now consider the equilibrium of segments cut across shear planes  $a-a$  and  $b-b$ , shown in Figs. 1–23c and 1–23d.



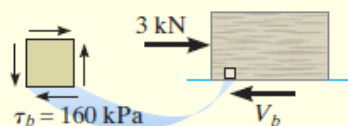
(c)

$$\pm \rightarrow \Sigma F_x = 0; \quad V_a - 3 \text{ kN} = 0 \quad V_a = 3 \text{ kN}$$

$$\pm \rightarrow \Sigma F_x = 0; \quad 3 \text{ kN} - V_b = 0 \quad V_b = 3 \text{ kN}$$

### Average Shear Stress.

$$(\tau_a)_{\text{avg}} = \frac{V_a}{A_a} = \frac{3(10^3) \text{ N}}{(0.1 \text{ m})(0.15 \text{ m})} = 200 \text{ kPa} \quad \text{Ans.}$$



(d)

$$(\tau_b)_{\text{avg}} = \frac{V_b}{A_b} = \frac{3(10^3) \text{ N}}{(0.125 \text{ m})(0.15 \text{ m})} = 160 \text{ kPa} \quad \text{Ans.}$$

The state of stress on elements located on sections  $a-a$  and  $b-b$  is shown in Figs. 1–23c and 1–23d, respectively.

## EXAMPLE 2.4

The plate shown in Fig. 2-7a is fixed connected along  $AB$  and held in the horizontal guides at its top and bottom,  $AD$  and  $BC$ . If its right side  $CD$  is given a uniform horizontal displacement of 2 mm, determine (a) the average normal strain along the diagonal  $AC$ , and (b) the shear strain at  $E$  relative to the  $x, y$  axes.

### SOLUTION

**Part (a).** When the plate is deformed, the diagonal  $AC$  becomes  $AC'$ , Fig. 2-7b. The length of diagonals  $AC$  and  $AC'$  can be found from the Pythagorean theorem. We have

$$AC = \sqrt{(0.150 \text{ m})^2 + (0.150 \text{ m})^2} = 0.21213 \text{ m}$$

$$AC' = \sqrt{(0.150 \text{ m})^2 + (0.152 \text{ m})^2} = 0.21355 \text{ m}$$

Therefore the average normal strain along the diagonal is

$$\begin{aligned} (\epsilon_{AC})_{\text{avg}} &= \frac{AC' - AC}{AC} = \frac{0.21355 \text{ m} - 0.21213 \text{ m}}{0.21213 \text{ m}} \\ &= 0.00669 \text{ mm/mm} \end{aligned}$$

*Ans.*

**Part (b).** To find the shear strain at  $E$  relative to the  $x$  and  $y$  axes, it is first necessary to find the angle  $\theta'$  after deformation, Fig. 2-7b. We have

$$\tan\left(\frac{\theta'}{2}\right) = \frac{76 \text{ mm}}{75 \text{ mm}}$$

$$\theta' = 90.759^\circ = \left(\frac{\pi}{180^\circ}\right)(90.759^\circ) = 1.58404 \text{ rad}$$

Applying Eq. 2-3, the shear strain at  $E$  is therefore

$$\gamma_{xy} = \frac{\pi}{2} - 1.58404 \text{ rad} = -0.0132 \text{ rad}$$

*Ans.*

The *negative sign* indicates that the angle  $\theta'$  is *greater than*  $90^\circ$ .

**NOTE:** If the  $x$  and  $y$  axes were horizontal and vertical at point  $E$ , then the  $90^\circ$  angle between these axes would not change due to the deformation, and so  $\gamma_{xy} = 0$  at point  $E$ .

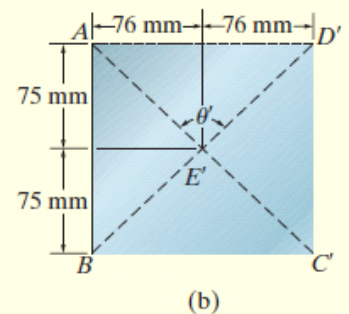
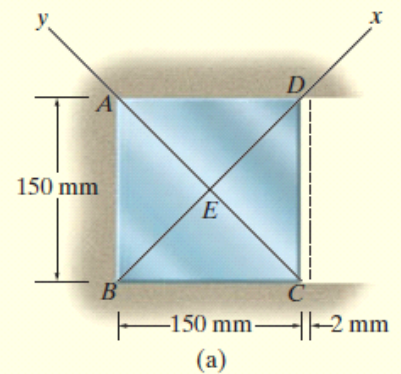
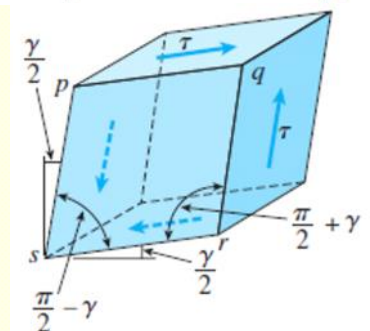
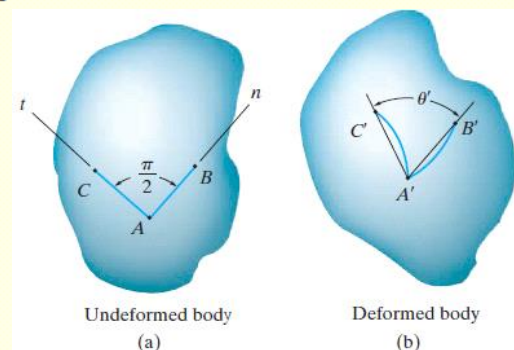


Fig. 2-7



## Ejercicios de repaso:

**Book: Mechanics of Materials**

**Author: R. C. Hibbeler**

**Edition: 8**

**Problems: 1-33, 1-34, 1-37, 1-38, 1-48, 1-51, 1-53, F2-4, 2-4, 2-6,  
2-19, 2-20**